National Center for Atmospheric Research

Electronic GPS Dropsonde System Installation

Electrical Loads Analysis Report: ELA-AVAPS-100

NSF Gulfstream V Aircraft, SN 677

Revision: A (initial release)

Date: 25 October 2005

National Center for Atmospheric Research Research Technology Facility 3450 Mitchell Lane Boulder, Colorado 80307

Prepared By:

Terry Hock Engineer National Center for Atmospheric Research

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Revision

Revision	Date	Description	Pages	Approved	Date
A	25 Oct 05	Initial Release	All		

NCAR Dropsonde System Electrical Loads Analysis NSF Gulfstream V Aircraft, SN 677

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1. Introduction

NCAR is developing main cabin equipment (MCE) racks to support atmospheric research equipment installations on the National Science Foundation (NSF) GV aircraft, SN677, under STC project number ST3285DE-T through the DOT-FAA Denver ACO. Research power wiring was previously installed by Lockheed Martin Aircraft Company IAW STC ST03056AT. Research equipment will not interface with existing aircraft systems, nor will the equipment provide information to the flight crew to assist them with operation of the aircraft. Therefore, AC 25-10 (Guidance for the Installation of Miscellaneous, Non-required Electrical Equipment) will be utilized to show compliance with the applicable regulations of 14 CFR Part 25. This report will present the data for the electrical load analysis of the Dropsonde Instrumentation rack.

This report is for the electrical load analysis for the Dropsonde System atmospheric instrumentation equipment installed in an instrumentation rack in the main cabin area. Items installed in the rack are Commercial Off The Shelf (COTS) equipment. There are four items which use aircraft power; 1) PC Rack Mount Computer, 2) LCD Flat Panel Monitor, 3) Telemetry chassis and 4) 28 volt Power Supply. All equipment in the Dropsonde instrumentation rack does not interface with existing aircraft systems, nor do they provide information to the flight crew to assist them with operation of the aircraft. Therefore, the miscellaneous, non-required electrical equipment and the guidance of AC 25-10 have been utilized.

2. Description

2.1. Research System

The research electrical system is designed to provide 40KVA of 115 VAC 400HZ 3 phase power continuously from the two Integrated Drive Generators (IDGs) for research applications. Twenty KVA from each IDG is allocated for Mission power. The aircraft uses frequency converters located in the Main Power Distribution Box (MPDB) to convert the 400 HZ to 21KVA of 110 VAC 60 HZ 1 Phase as supplied by the manufacturer. Research power distribution and controls are located in the MPDB as supplied by the manufacturer. The maximum allowable continuous electrical load for each power source is 20 amps.

2.2. Research Equipment

The Dropsonde system consists of the following equipment:

- Dropsonde instrumentation rack using 115 VAC 60 Hz @14.7 amps. The unit is protected by the existing CB#3303CD11, a 20 amp breaker located in the MPDB. Table 1 lists the individual components contained in the dropsonde instrument rack. Several of the units are protected by an independent circuit breakers, see table 2.
- Dropwindsonde Launcher Assembly, manufactured by Aeromet/L3 Communications Inc., model number 01901-10101-501. The unit is powered by 28 Vdc from the Dropsonde instrumentation rack. The wiring

harness between the instrumentation rack and launch tube is protected by a DC 20 amp circuit breaker located in the Dropsonde instrument rack.

• UHF Aircraft blade antenna, Manufactured by Sensor Systems Inc., model number S65-1217. This is a passive device and does not consume power.

The Dropsonde research instrumentation rack consists of the following commercial off the shelf (COTS) equipment: listed in Table 1

Item	Description	Manufacture	Model	Input Voltage
1	PC Computer	PCS	2USCP43	120VAC 60Hz
2	LCD Monitor	Interlogic Industries	RDF19AX-SHB	120VAC 60Hz
3	Telemetry Chassis/Cooling Fan	NCAR	AVAPS Chassis	120VAC 60Hz
4	28Vdc Power Supply	Vicor	VP-F1311991	120VAC 60Hz
5	Launch tube control Panel	NCAR	Dwg No 67705- AVAPS-1-3	28 Vdc
6	Keyboard	Cyber Research	OIX 1910B-P	NA
7	Sonde Storage box	NCAR	Dwg No. 67705- AVAPS-1-3	NA

Table 1 Dropsonde Equipment List

A block diagram (Dwg. No. AVAPS-100-01) of the AC and DC power system is included at the end of the report of the Dropsonde installation.

Table 2 Circuit Breaker Data

ltem	Description	Circuit Breaker Rating	Mill-Spec Part No.
1	Cooling Fan	2 amps	MS3320-2
3	Telemetry Chassis	1 amps	MS3320-1
4	Vicor Power Supply	5 amps	MS3320-5

2.3. Circuit Protection

The equipment rack installed is individually protected by circuit breakers or by a shared breaker installed by the manufacturer in the MPDB, the breaker shall not exceed 20 Amps. The circuit breakers are of the same type as installed by the manufacturer as original equipment.

2.4. Load Analysis

Below in table 1 is a list of all devices in the Dropsonde Instrumentation rack that is power by the aircraft 110 VAC 60Hz power system. The equipment has built in circuit breaker protection for each individual item listed in table 1. The maximum total current load under any condition is 11.7 amps.

	Tabulation of power values AC Power 120VAC @ 60 Hz for all units listed										
ltem	Description	AC Power Cable Type	Average Current (Amps)	Peak Current (Amps)	Power Not to Exceed (Watts)	Unit Circuit Breaker (Amps)					
1	Computer	3-wire 16 gauge	0.8	1.3	350	None					
2	Monitor	3-wire 16gauge	0.4	0.4	50	None					
3	Telemetry Chassis/ Cooling Fan	3-wire 16 gauge	0.5	0.7	100	2					
4	Vicor Power Supply	3-wire 16 gauge	0.3	3.5	500	5					
	Total		2.0 Amps	5.9 Amps	1000 Watts	7 Amps					

Table 3 Power Load Summary

The maximum electrical load current at 120VAC 60HZ of the Dropsonde Instrument rack is 5.9 amps. This load allows for a safety margin of 14.1 amps (or 70.5 %) from the 20 amp circuit for the MCE.

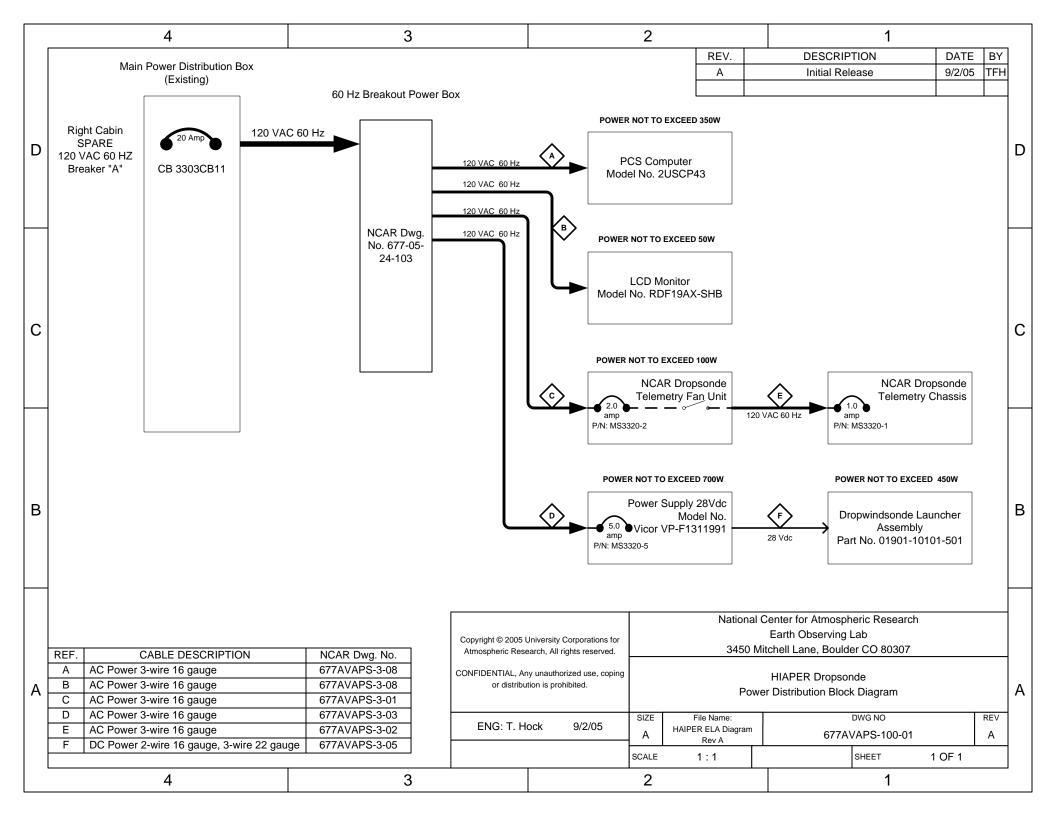
3. Conclusion

The basic research rack system as installed by NCAR is a stand alone independent system that interfaces to the Lockheed Martin Aircraft Center as part of STC number ST03056AT research electrical system. This research power distribution system incorporates automatic load shedding and a cockpit kill switch for the research power system. Independent circuit breakers protect all of the rack installed equipment and are of the type supplied as original equipment by the manufacturer. The system loads cannot exceed 20 amps per circuit.

All the circuit breakers are accessible to the flight crew at the Main Power Distribution Box.

The basic research systems do not interface with existing aircraft systems, nor do they provide information to the flight crew to assist them with operation of the aircraft. Therefore, all basic research system equipment can be considered miscellaneous, non-required electrical equipment and the guidance of AC 25-10 has been utilized.

Figure 1 Block diagram of power distribution for HIAPER Dropsonde Instrument



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Make	Model No.	KAFT COMPONENT IDENTI	ICATION	
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