

to hear the audio in the Adobe Reader, you must select the options button at the start and "trust" this presentation

# Measuring State Parameters of the Atmosphere

## Some Applications of Atmospheric Thermodynamics

Al Cooper

Earth Observing Laboratory, NCAR

IDEAS-4 Tutorial

# Introduction

for more information  
on navigation tools,  
wait for audio to end  
and click this box

## Goals of This Presentation

Present two complementary aspects related to atmospheric thermodynamics:

- 1 Discuss some basics regarding how measurements of thermodynamic state variables are measured by a research aircraft
- 2 Show some useful applications of atmospheric thermodynamics to how those measurements are made



# STATE VARIABLES

## What Are State Variables?

- Those variables needed to specify the thermodynamic state of the system, in this case the atmosphere.
- If we consider a moist atmosphere, in general we need three variables to specify the state. They may be taken, for example, to be *temperature*, *pressure*, and *water vapor pressure*.

Other variables can then be determined from these, for example:

- **density** from the perfect gas law
- **relative humidity** from knowledge of the equilibrium vapor pressure vs  $T$  for water
- **dew point** also from knowledge of the equilibrium vapor pressure vs  $T$  for water

did you catch my mistake? Click here  
when the audio ends for info.



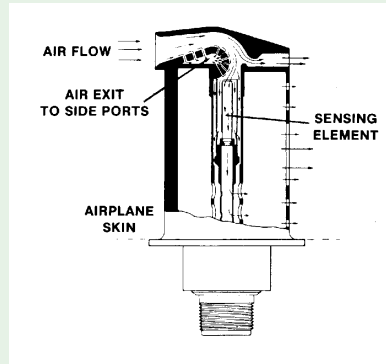
# TEMPERATURE SENSORS

## Types of Temperature Sensors

- 1 Resistive-element sensors:
  - often, platinum wire
  - resistance varies with temperature
- 2 radiometric:
  - CO<sub>2</sub> absorbs and re-emits radiation in short distances at some specific wavelengths
  - The intensity of such radiation varies with the temperature
- 3 Others sensors are also sometimes used, including thermocouple junctions and thermistors



# THE STANDARD SENSOR



# EFFECT OF AIRSPEED ON IN SITU SENSORS

## Airflow Approaching a Stagnation Point



- At boundaries, airspeed tends to zero relative to the boundary.
- The result is compressional adiabatic heating of the air
- The sensing wire therefore is in contact with warmed air, not ambient air

# WHAT IS TTX?

## Definition

TTX is the measured temperature determined from the resistance of the wire.

- It is named “total” temperature because it is approximately the total temperature of air brought to a stagnation point.
- It is actually closer to the “recovery” temperature, defined below



# HOW IS TTX RELATED TO ATX?

## Conservation of Energy

First Law:  $dU = \delta Q - \delta W$ , and for a perfect gas  $dU = c_v dT$

On a streamline starting with varying speed  $V$ ,

$$\delta Q = 0 \text{ and } -\delta W = p\delta V$$

To  $U$ , must add kinetic energy  $\frac{1}{2}\rho V_a^2$ , with air density  $\rho_a$ :

$$\frac{1}{2}V^2 + c_v T + \frac{p}{\rho_a} = \text{Constant}$$

Because  $\frac{p}{\rho_a} = R_d T$  and  $c_v + R_d = c_p$ ,

$$\frac{1}{2}V^2 + c_p T = \text{Constant}$$



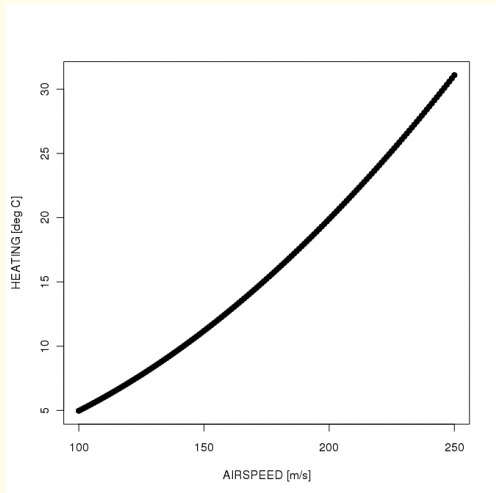


# MAGNITUDE OF THE HEATING

Heating is about  $5^{\circ}\text{C}$  at 100 m/s, increasing to about  $30^{\circ}\text{C}$  at airspeeds reached by the GV.

Accurate correction for this airspeed thus is quite important, especially at high airspeed.

► Skip Details: The Recovery Factor



# THE RECOVERY FACTOR

## A Further Correction That Depends On Probe Geometry

- The air does not reach complete stagnation at a distance in thermal contact with the sensing wire. One might expect that the temperature that affects the wire is that present at a distance from the sensor of about a mean free path for air molecules.
- This is usually dealt with via a “recovery factor” that varies with sensor but may be as high as 0.98 (where 1.0 would apply for a stagnation point).
- Often this is determined from flight maneuvers where the aircraft varies airspeed while flying through a region of uniform temperature so the effect of airspeed on the measurement can be detected.

want still more details?



# WHAT HAPPENS IN CLOUD?

If the sensor becomes wet, the measurement will be wrong

- Consider:  $RH \approx 100\%$ , wet sensor
- Air is heated on approach to sensor, so RH decreases
- The water on the sensor and in drops partially evaporates, cooling the sensor
- At the extreme, the sensor approaches the wet-bulb temperature, cooling by a few degrees Celsius

▶ [Skip Details on the wet-bulb temperature](#)



# THE WET-BULB TEMPERATURE

## Basic Formula

$$rL_V + c_p T = \text{Constant}$$

$r$  is the water-vapor mixing ratio,  $L_V$  the latent heat of vaporization,  $c_p$  the specific heat at constant pressure, and  $T$  the temperature.

- As the air evaporates from the sensing wire and cloud drops,  $T$  decreases and  $r$  increases.
- At saturation,  $r = r_s(T_{WB})$ . The wet-bulb temperature is the temperature for which this condition is satisfied.
- Conceptually, one could plot the quantity  $r_s(T')L_V + c_p T'$  vs  $T'$  and find the point at which that curve intersects the value specified by the formula above for ambient conditions  $\{r, T\}$ . In practice, the equation is usually solved iteratively.



# HUMIDITY SENSORS

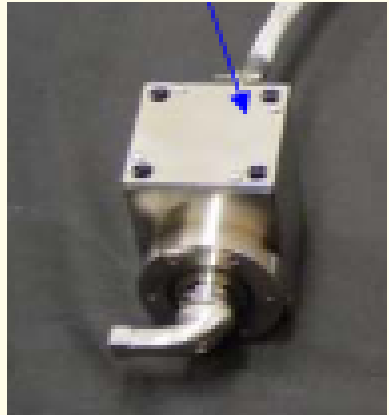
## Basic Sensor Types:

- **Dew point hygrometers:** Devices that detect the presence of condensate on a chilled mirror
- **Light-absorption hygrometers:** Devices that measure the absorption of radiation at a wavelength where there is strong water absorption
- **Wet-bulb thermometers:** Devices that measure the cooling of a wetted sensor
- **Capacitance measurements or hygristor (resistance) devices:** Common in radiosondes, seldom used in research aircraft



# A CHILLED-MIRROR HYGROMETER

The sensor housing



# A CHILLED-MIRROR HYGROMETER

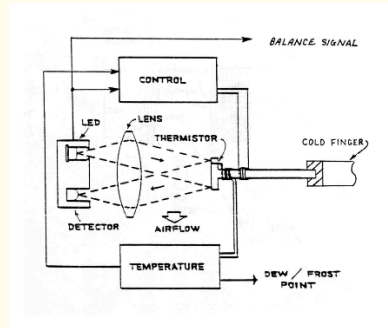
Photograph as mounted on the  
GV



# A CHILLED-MIRROR HYGROMETER

## The control process

- Reflected light from the mirror is measured
- If the reflected light decreases, the mirror is heated, and v.v.
- The control loop is adjusted to keep just threshold condensation on the mirror



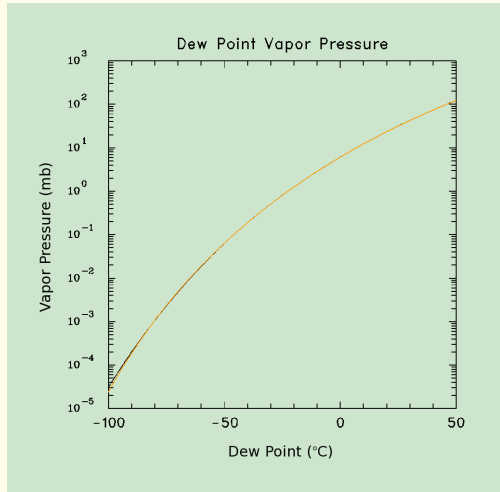


# USING CHILLED-MIRROR MEASUREMENTS

## Finding the Water Vapor Pressure

### Definition

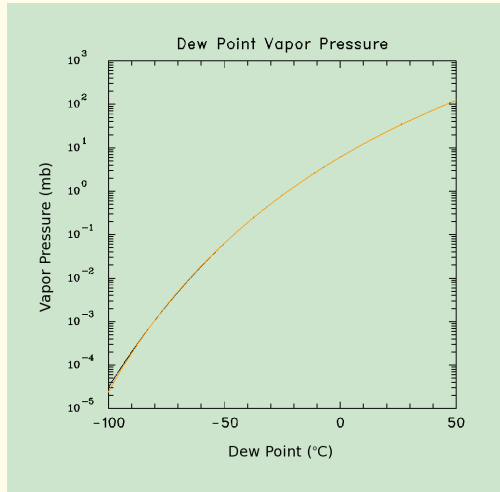
The dew point is the temperature at which the water vapor pressure would be in equilibrium with a plane water surface.



# USING CHILLED-MIRROR MEASUREMENTS

## Finding the Water Vapor Pressure

The functional dependence is usually expressed as  $e = e_s(T_{DP})$  where  $e_s$  is the equilibrium vapor pressure function,  $e$  is the vapor pressure, and  $T_{DP}$  is the dew point.

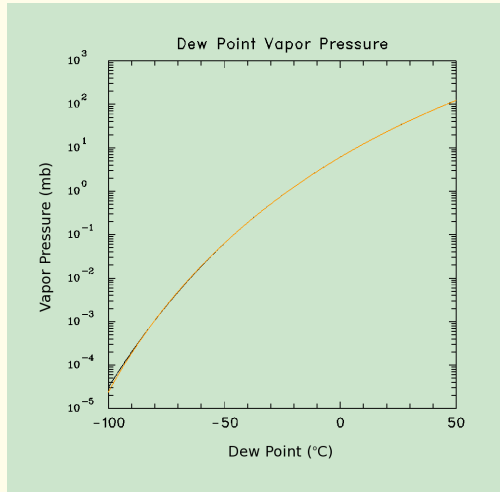


how do we know this for low T?

# USING CHILLED-MIRROR MEASUREMENTS

## Finding the Water Vapor Pressure

Formulas exist to express the function  $e_s$ , including the Clausius-Clapeyron equation, the Goff-Gratch formula, or the Murphy-Koop formula. We now use the latter.



## FURTHER CONSIDERATIONS

Three additional considerations when using these formulas:

- 1 The condensate on the mirror may be frost, not dew.
- 2 The pressure in the sensing chamber may differ from the ambient.
- 3 In the presence of dry air, the equilibrium vapor pressure over a plane surface is slightly higher than the equilibrium value in the absence of air.

All these require corrections. The first two are often substantial.



## HUMIDITY SENSORS THAT MEASURE ABSORPTION

## Beer's Law

$$\frac{dl}{l} = -\sigma n dl$$

[ $l$  is the intensity of radiation,  $\ell$  distance,  $\sigma$  the molecular cross-section for absorption, and  $n$  the number density of molecules.]

$$I = I_0 e^{-\sigma \ell n}$$

- The quantity measured by instruments using absorption is then  $n$  or  $\rho_w = m_w n$ , the mass density of water vapor.
- Two types of radiometric hygrometer are in common use:
  - 1 Lyman-alpha hygrometers based on absorption of the Lyman-alpha line of hydrogen, which lies in the UV
  - 2 Tunable diode laser (TDL) hygrometers that work in the near IR



# THE HUMIDITY VARIABLES

- Original Measurements: **DPB**, **DPT** (mirror temperatures)
- Corrected for frost-dew difference, etc: **DPBC**, **DPTC**, **DPXC**
- Derived:
  - EDPC** water vapor pressure
  - MR** water vapor mixing ratio
  - RHUM** relative humidity
  - RHODT** water vapor density
- Experimental: **MIRRORT\_CR2** (cryogenic hygrometer) and some measurements from a TDL hygrometer. These are not yet processed to final engineering-unit form.



# CALCULATING THE DERIVED VARIABLES

## Preferred dew-point sensor

One of the dew point measurements is designated as the preferred measurement; e.g., DPXC=DPTC. Derived measurements are determined from this basic measurement.

- EDPC: determined from  $e=e_s(T_{DP})$  after corrections as discussed earlier
- RHUM:  $e/e_s(T)$  where  $T$  is the ambient temperature
- MR:  $r = \frac{\varepsilon e}{p-e}$  where  $\varepsilon = M_W/M_a$  is the ratio of the molecular weight of water to that of air and  $p$  is the total pressure
- RHODT:  $\rho_w = \frac{e}{R_w T}$  where  $R_w$  is the gas constant for water vapor. This equation is also used to obtain  $e$  from measurements of  $\rho_w$  or  $n$ , such as provided by the radiometric hygrometers.



# PRESSURE MEASUREMENT

## The Sensors

- Many different transducers are available to measure absolute or differential pressure.
- Among the most accurate are digital quartz crystal sensors that change oscillation frequency with pressure.
- Others are capacitative, piezoelectric, piezoresistive, ...
- On aircraft, these attach to static ports designed to provide pressure close to the flight-level pressure





# HOW DO STATIC PORTS WORK?

## The Key Problem

Airflow around the surfaces of the aircraft creates a varying pressure field that makes accurate measurement difficult.

- Example from earlier discussion of temperature measurement:

$$\Delta p = \rho_a \frac{V^2}{2}$$

$\Delta p$  can be 70 hPa under the following conditions:  $p = 200$  hPa,  $T = -40^\circ\text{C}$ ,  $V = 220$  m/s.

- Pressure ports: “static buttons” located at special locations where this effect is minimized. (Corrections are still necessary.)
- Calibration: “trailing cone” and flight maneuvers to test the effects of angle of attack and sideslip

## MAPPING PRESSURE FIELDS

## Heights on a constant-pressure surface show pressure gradients

- GPS measurements give the height of the aircraft to few-cm accuracy. (Synoptic maps of pressure fields often use contour increments of 50 m or more.)
- It is possible to map mesoscale pressure fields by flying on constant-pressure surfaces.
- Accuracy considerations: If uncertainty in  $p$  is 0.5 mb, then the corresponding uncertainty in height can be estimated from the hydrostatic equation:

$$\frac{dp}{p} = -\frac{g}{RT}dZ$$

For  $\delta p = 0.5$  mb,  $p = 500$  mb,  $T = 263$  K, gives  $\delta Z = 7.5$  m.



## More Information:

### Contact Information:

email: [cooperw@ucar.edu](mailto:cooperw@ucar.edu)

phone: 303 497 1600

